

— Survey Grade INS

Use in airborne applications

Operating handbook



Document
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This operating handbook aims to guide users during sensor installation and configuration in airborne environments.

You don't need the sbgCenter to configure the products, only the web interface is required.

Mechanical installation

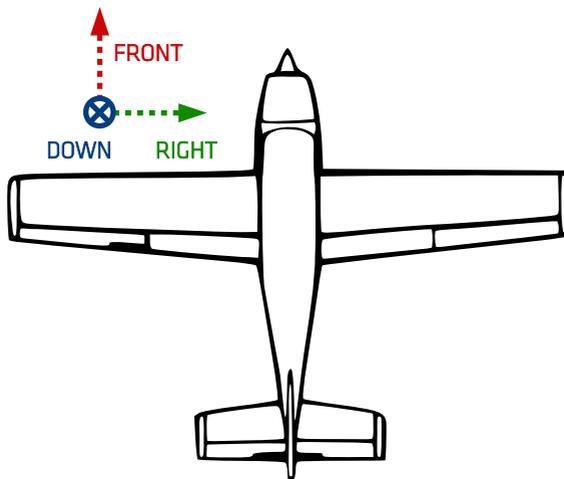
When used in airborne application, the INS performs some velocity assumptions depending on the motion profile (Airplane or Helicopter). You still need a good sensor installation to get best results, especially for high dynamic applications.

For an optimal installation, please make sure these few **critical** points are respected:

- Sensor is **rigidly fixed** to the frame
- Sensor is **not moving** in regard to other equipments (antennas, LIDAR, etc...)
- Sensor should be protected from high temperature gradients

Aircraft Reference Frame

The aircraft coordinate frame is defined as follows:



Note: For the Navsight version, names “sensor” and “IMU” in this document refer to Ekinox-I, Apogee-I, or Horizon-I units (IMU block only), distinctly from the Navsight Processing unit (or P.U).

Sensor orientation in the aircraft

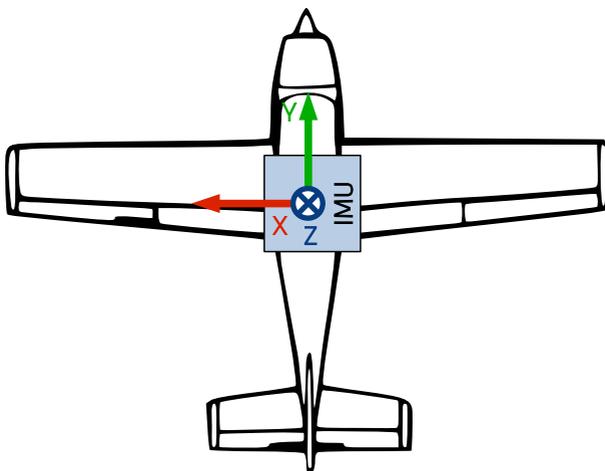
The sensor can be placed in any orientation in the aircraft. When IMU axes do not match exactly with the aircraft coordinate frame, a two step software alignment method can be performed, as explained below.



Note: Once the unit is aligned (mechanically or by software), the aircraft reference frame is used for all lever-arms in the Inertial Unit configuration.

Step 1: Rough Misalignment

In the first step, you basically need to check in which direction each IMU axes point. Next example shows IMU X axis is turned left, IMU Y axis is turned in aircraft Front direction and Z is turned down.



Step 2: Fine misalignment (Optional)

This step can be used to compensate for small angles in case the sensor is not exactly aligned with the aircraft coordinate frame. The angles are expressed in terms of Euler Roll, Pitch and Yaw residuals.

These residuals can be measured by using optical, or multi-antenna GNSS systems. For instance, you can leave the aircraft horizontal (where you should expect a zero roll and pitch angles) and average the unit roll and pitch measured angles. These averaged values should directly be used as misalignment angles.

Sensor location and monitoring points

Primary Lever Arm (optional)

You can place the sensor anywhere on the aircraft, however it should be placed at a reasonable distance **from the aircraft Center of Rotation** (avoid placing the sensor at the wing edge).

The primary Lever Arm can be measured to provide an additional output at this specific location. It is the signed distance, in the aircraft coordinate frame, FROM the IMU TO the Center of Rotation.

Output Monitoring points

The unit can produce all the navigation data and heave outputs at up to 5 location:

- The AHRS / INS
- The Center of Rotations
- The 3 user defined Monitoring Points

To define a user monitoring point, the lever arm from the IMU to the monitoring point should be measured, just as the Primary lever arm.

Please remember all lever arms are measured in the aircraft coordinate frame (not IMU coordinate frame).

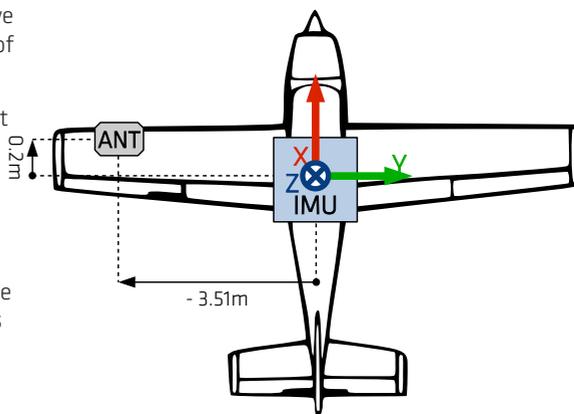
GNSS Antenna placement

Single antenna installation

Single antenna installation is possible, but not recommended in medium to low dynamics applications, because of a lower heading observability.

Main installation requirement is to have the GNSS antenna with a clear view of sky.

Once installed, the GNSS lever arm must be measured. It is the signed distance, expressed **in the aircraft coordinate frame**, FROM the sensor center of measurements, TO the main GNSS antenna. It must be measured **within 5cm accuracy**. A calibration can be performed to estimate these lever arms within 1 cm of accuracy.



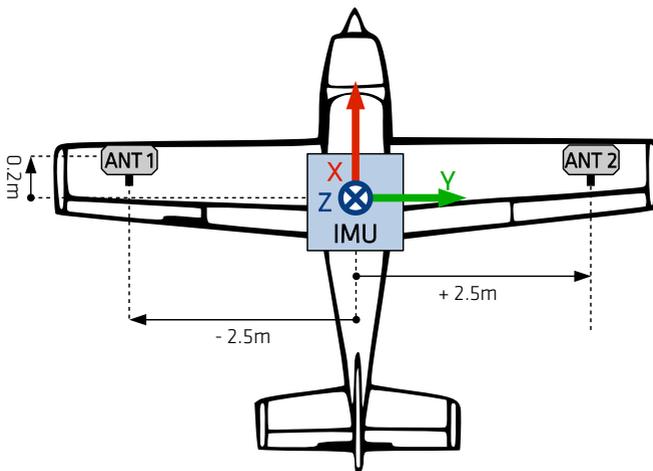
Dual antenna installation

When installing a dual antenna setup, the following requirements should be respected:

- The antennas should be **fixed** in regard to the inertial unit
- Both antennas must be **turned the same way** (connectors oriented in same direction)
- Both antennas must have the **same optimal view of sky**.
- **Same cables** with same length must be used for both antennas. Prefer no splitter, or make sure they are adapted.
- Baseline of **at least 3 meters between both antennas** for best performance

Once installed, the GNSS lever arms must be measured. These is the signed distances, expressed **in the aircraft coordinate frame**, **FROM** the IMU center of measurements, **TO** each GNSS antenna. These must be measured **within 5cm accuracy**. A calibration can be performed to estimate these lever arms within 1 cm of accuracy.

The following diagrams show a typical dual antenna installation, with antenna 1 (position and velocity) at the left and antenna 2 (heading) on the right wing.



Software configuration

All the configuration is done through the web interface.



Note: At the first access or if the device firmware has been updated, the sensor will cache the entire embedded website to optimize the responsiveness. This preload operation may take up to one minute depending on your system configuration.

IMU Setup (Navsight series only)

Select the IMU you'll be operating with your Navsight processing unit, both Ekinox and Apogee ranges are available, in surface or sub-sea housing:

Device Settings
✕

- ☰ Installation Overview
- 📡 IMU Setup
- 📡 Sensor
- 📡 Aiding Assignment
- 📡 Aiding Setting
- 📡 Inputs/Outputs
- 📡 Data Output
- ⚙️ Advanced
- 👤 Administration

IMU Model Selection

Please select the IMU model that is connected to your NAVSIGHT processing unit.



EKINOX2 Surface



EKINOX2 Subsea



APOGEE Surface



APOGEE Subsea



HORIZON Surface

APOGEE Subsea

The APOGEE is a high performance and ITAR free MEMS based Inertial Measurement Unit able to provide better than 0.008 rollpitch accuracy. The subsea housing is rated to 200 meters and is ideally suited to be pole mounted for multi-beam survey applications.

IMU Measurement Point

You can select which point to use as a reference for all mechanical installation measurements.
 The "Bare IMU" mode doesn't apply any offset on physical IMU data.
 If you select "Cover Target", the (0,0,0) point will be set to the center of the frame reference drawn on top of the IMU cover.

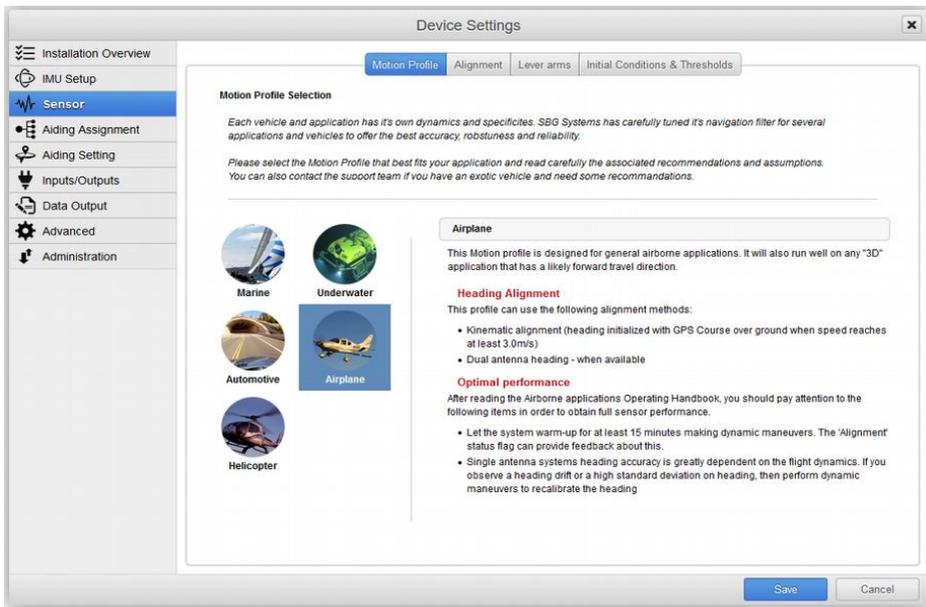
Select the IMU reference point to use: Bare IMU

Save
Cancel

Sensor

Motion profile

Conventional aircraft should use the “Airplane” motion profile, for helicopter or vertical landing and take off applications you should use the “Helicopter” motion profile.

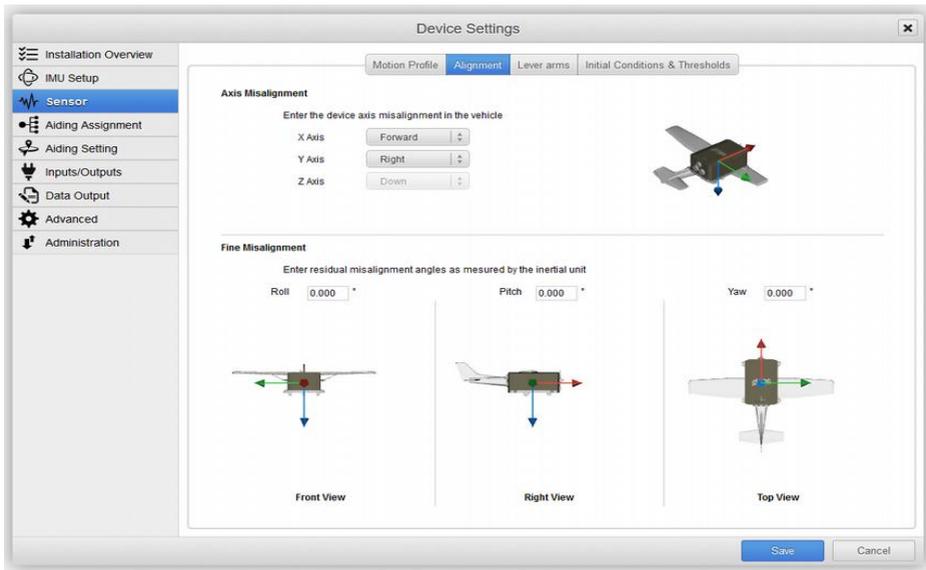


The “Helicopter” profile will use GNSS + Accelerations by default for the heading initialization and will have no specific assumption on motion. The “Airplane” profile will rely on the GNSS Course by default and will assume the typical motion direction is forward.

The “Airplane” motion profile will also assume there will higher velocities and high dynamics.

Misalignment

Here you can configure the device misalignment in the vessel coordinate frame. The nice looking view will help you through these steps.



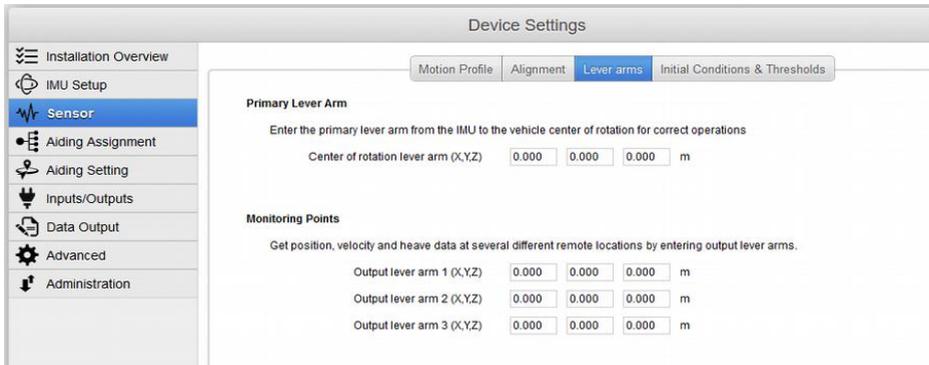
Axis Misalignment

First, you need to enter the axis misalignment settings, by entering the sensor X and Y axes direction in the aircraft frame.

Fine Misalignment (Optional)

In a second time, you can enter the residual roll, pitch and yaw angles if the sensor is not fully aligned with the aircraft frame after the Axis misalignment step

Lever Arms (Optional)



Primary Lever Arm

Enter here the lever arm **from** the sensor **to** the center of rotation in Aircraft Reference Frame,

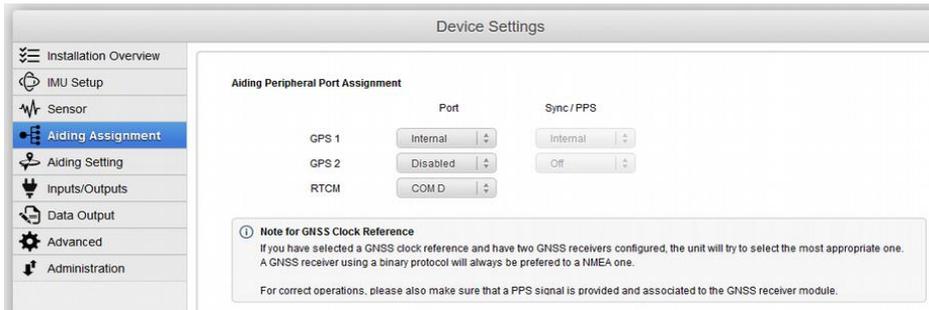
Monitoring Points

You can configure three monitoring points to output all orientation and position data.

 **Note:** These values can also be changed directly in the 3D Setup Overview section.

Aiding Assignment

You can enable one or two GNSS on this panel, you can chose whether you want to use the internal GNSS or not. The RTCM correction input has to be selected here for RTK applications.



Note: If an external GNSS receiver is used, make sure to select the source of the PPS signal on the right input synchronization. In case of an internal GNSS receiver, you can activate here the RTCM reception for RTK operations.

GNSS Configuration

Please check the following points to ensure a correct GNSS configuration:



GNSS Setup

Choose this parameter for external receiver, otherwise keep “Internal”. Depending on the GNSS you are using (NMEA, Novatel, Trimble or Septentrio), you can refer to the corresponding manual to know how to configure the GNSS.

If two antennas are used you have to define the lever arm for the second one. But you can also choose “auto lever arms” in GNSS Heading Mode to let the Kalman Filter estimate the antennas lever arms through a specific calibration run.



Note: When using Dual Antenna (Auto Lever Arm), the heading will not be calculated until the calibration is performed. The secondary antenna lever arms are necessary to calculate the GNSS heading.

GNSS Lever arms

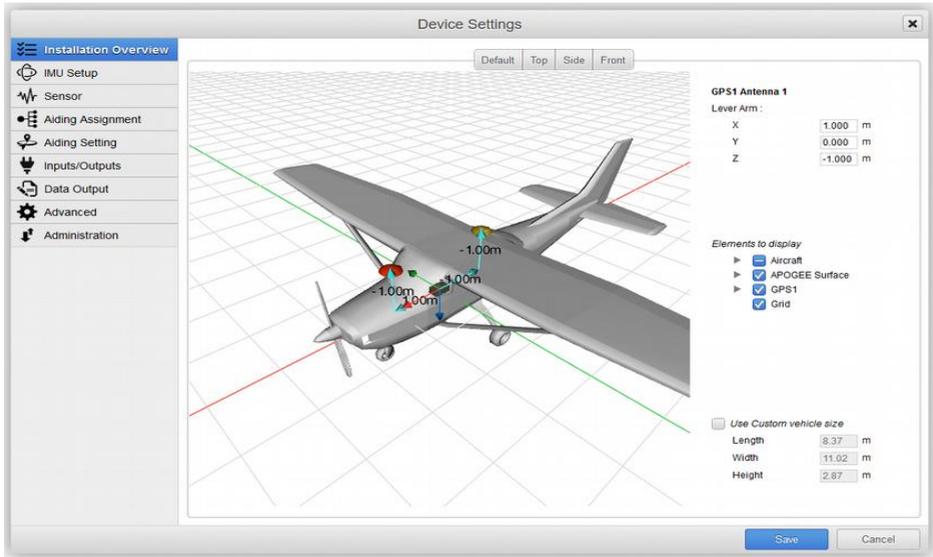
Set up the lever arm of the GNSS depending on its position on the aircraft (GNSS Antenna placement).

Aiding Use and Rejection

Auto rejection is advised for all measurements. This mode automatically detects if a measurement can be trusted or not.

Installation Check

The whole configuration can easily be checked in the Setup Overview section. It features a 3D display that can be modified by simply clicking on an item, then changing the orientation or lever arm values. Any change will be directly visible.



GNSS lever arm and alignment calibration procedure

The Ekinox, Apogee, Navsight and Quanta series include a very intuitive and accurate calibration procedure that simplifies a lot the installation while allowing highest performance level.

As a rule of thumb, measuring GNSS antenna lever arms can be very complex in typical situations where the IMU is installed inside the vehicle cabin and the GNSS antennas are located outside. Moreover, a dual antenna setup is even more complex due to the fact that a good knowledge of the mechanical alignment is required to allow proper measurement of the lever arms.

Considering this challenge, the automated calibration procedure developed allows the user to:

- Enter rough GNSS lever arm for the primary GNSS antenna – within 5 – 10cm
- If the vehicle dynamics are sufficient, let the system determine automatically the secondary antenna lever arm – nothing to enter
- In case of lower dynamics, enter a rough estimation of secondary lever arm – within 5 – 10cm.
- Run a one time calibration procedure to refine the GNSS lever arms
- Store parameters in the non-volatile memory to obtain highly accurate measurements in following missions.



Note: For a 20 minute calibration, RTK and dynamics are required. Low dynamic or non RTK applications can take longer to perform a calibration (1 hour).

Step by step calibration procedure

Step 1: define rough GNSS lever arms in configuration page

In the GNSS aiding settings, you have the possibility to select the GNSS heading mode between three options:

- Single antenna mode
- Dual antenna mode (auto lever arm)
- Dual antenna mode (known lever arm)

GNSS Setup

Select the receiver model and if you plan to use single or dual antenna mode. Dual antenna heading is useful for low dynamics applications and to initialize the INS in static conditions.

Receiver Model GNSS Heading Mode

GNSS Lever Arms

Please enter the primary lever arm FROM the GNSS antenna TO the INS with an accuracy better than 20 cms. Before the INS can use dual antenna heading, you will have to perform a successful calibration.

Primary Antenna (X,Y,Z) m

Figure 1: GNSS aiding configuration page

In the first two options, only the primary GNSS antenna lever arm can be entered. The third option requests a user first guess for the secondary antenna lever arm.

Step 2: Start the calibration

Before starting the calibration, it is recommended to place the vehicle in a good GNSS environment to enable best performance. Although it is not required, we recommend the calibration procedure to be performed with high precision GNSS like PPP or RTK as it will provide faster results, with a better confidence.

From the web page, in the calibration tab, you can simply start the calibration by pressing “Start Calibration” or “Restart Calibration” button.

This will cause the Extended Kalman filter to restart in a specific mode that enables estimation of GNSS parameters.

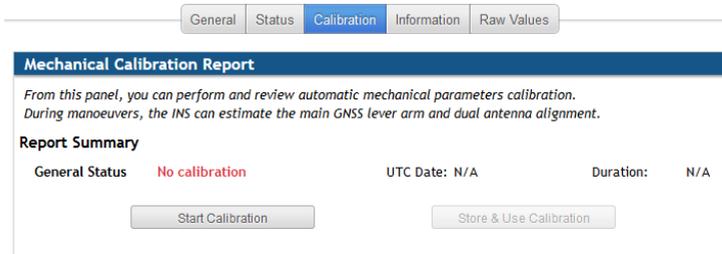


Figure 2: Default calibration page with no calibration applied

Step 3: Running calibration and check progress

Once started, the calibration status will typically do into “Waiting” state. In order to actually run the calibration, we need to operate the Kalman filter in Full navigation mode, which means we need heading, and position resolved.

In case the setting “Auto lever arm” has been set, please note that the system will not be able to initialize the heading in static condition, so the EKF will transition to Full navigation only after a short period of motion in forward direction.

Once the calibration is started and the vehicle is operated a sufficient speed (higher than 2.5m/s), the calibration status will transition to “running” mode. Two progress bars now display the calibration progress: one for the primary GNSS lever arm estimation, and one for the dual antenna heading alignment (linked to the secondary lever arm). The more dynamics we can get, the faster the calibration will be.

The typical recommendation is to perform high speed maneuvers, eight shape patterns, accelerations and deceleration phases.

To get more advanced feedback on the performance of estimated parameters, the calibration page also displays the estimated lever arms and angles, in comparison to what you entered initially, with associated standard deviations.

Depending on the GNSS environment and precision (RTK or not) and vehicle dynamics, the calibration can be performed within a few minutes, or can take more than half an hour. In case of low dynamics it may be impossible to reach a 100% finished calibration

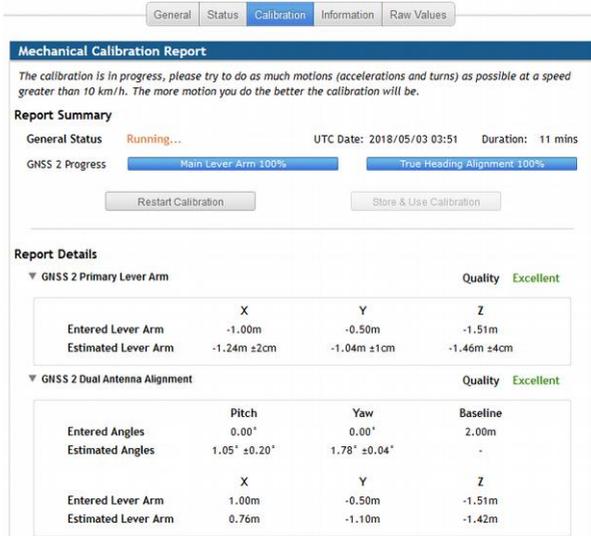


Figure 3: Running calibration in known lever arm mode

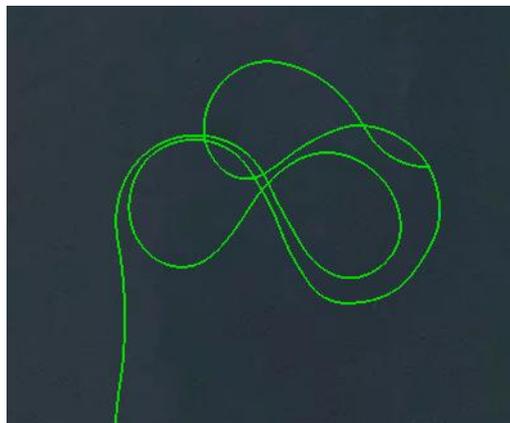


Figure 4: Typical maneuvers during a calibration

Step 4: Ending calibration

When active, the calibration continuously improves the lever arm and alignments. Even after reaching 100% completion, it is still possible to enhance the estimated values again by keeping maneuvering.

On the opposite side, in case of poor GNSS environment and/or low dynamics, it might be challenging/impossible to reach a 100% complete calibration.

That’s why in the end, it belongs to the user to decide when the calibration should be stopped. User should also verify the consistency of the estimated parameters with respect to the entered values and actual setup. A minimum of 20% completion on each estimated parameter is required to unlock the calibration ending.

Once the calibration results are satisfactory, user can click on “Store & Use Calibration” button. This action will not restart the Kalman filter and he can move on to the actual mission directly. At next start, the unit will automatically load calibrated lever arms.

In case of inconsistency between actual setup and estimated parameters, user has the possibility to:

- Try a new calibration from scratch by pressing “Restart Calibration”
- Enter rough estimate of secondary GNSS lever arm to help the calibration procedure; then restart the calibration.

Checking what is actually applied, clearing calibration data

In any case, the parameters applied in the navigation filter are the one displayed in the configuration panel that can be freely edited. In case the setup is changed – compared to the previously applied calibration – it is always possible to modify the settings and start a new calibration, based on the newly entered values.

The calibration page always reflect the report from last calibration performed. In the same time, it checks whether this calibration is consistent with currently applied settings, using status “Applied and used” or “Not used”.

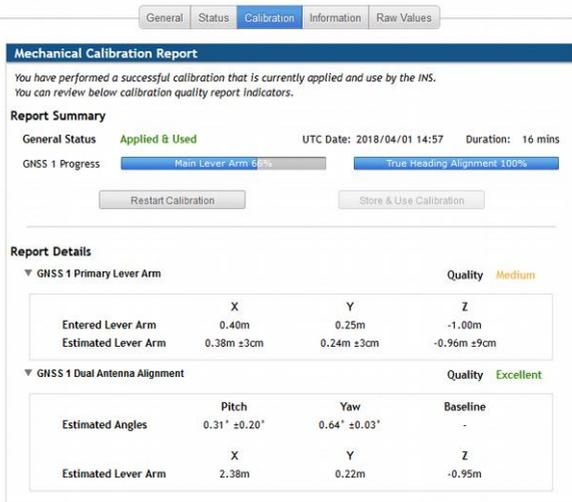


Figure 5: Applied calibration in auto lever arm mode

Operation

Heading alignment (initialization)

At power up, the unit is able to provide roll and pitch angles. Full navigation data becomes available once GNSS has a correct fix, and the system could initialize the heading angle.

Depending on the type of application, and the GNSS setup, there can be several heading alignment methods available:

Dual antenna alignments

In case of a dual antenna GNSS setup, heading alignment can be performed in static. Unit should be started with a clear view of sky to prevent bad initialization of the GNSS true heading.

Single antenna, Airplane mode

In case of a single antenna installation, and when using a conventional airplane, the heading will initialize as soon as the aircraft is moving at a higher speed than 10km/h.

Single antenna, Helicopter mode

On helicopters, with single antenna setup, the heading will initialize as soon as the helicopter is (horizontally) accelerated at a higher rate than 3m/s^2 . Note this mode also requires a clear view of sky for proper operation.

Warm-up

For all applications, full performance is reached after a warm-up time of 5 minutes. The system is operational before that time, but performance parameters cannot be guaranteed.

Ideally, the beginning of a mission should include some motion patterns with GNSS availability that will be used to let the Kalman filter converge. This is the alignment phase.

There is no mandatory pattern to perform, the sensor will only need as much dynamics as possible (orientations and accelerations). A typical alignment pattern is shown in the next picture:



A few “eight” figures are sufficient most of the time, and the reported status flags will help you to validate that the alignment phase has ended.

Status check

The web page provides advanced status feedback to make sure the system is working properly.

The Alignment Status is switched to green when internal filter parameters have converged and the system can achieve optimum accuracy. Other quality indicators focus on the accuracy of individual outputs (orientation, position, velocity).

Solution	
Solution mode	Nav position
Alignment status	Aligned
Quality	
Attitude	✓
Heading	✓
Velocity	✓
Position	✓

Once these status flags are green, you can start your survey!

Note: These status flags turn green when they reach a certain threshold. These thresholds can be configured in the section below:

- ☰ Setup Overview
- Sensor**
- Aiding Assignment
- Aiding Setting
- Inputs/Outputs
- Data Output
- Advanced
- Administration

Motion Profile | Alignment | Lever arms | **Initial Conditions & Thresholds**

Initial Conditions

Initial position might be used to compute local gravity and local magnetic declination before GPS becomes available.

Date: Leap Seconds:

Latitude: ° ' '' Longitude: ° ' '' Altitude: m above ellipsoid

Validity Thresholds

You can define your own validity thresholds for each type of measurement. These thresholds are compared to the Kalman Filter estimated standard deviations. They are used to trigger validity flags on the web interface and for output logs.

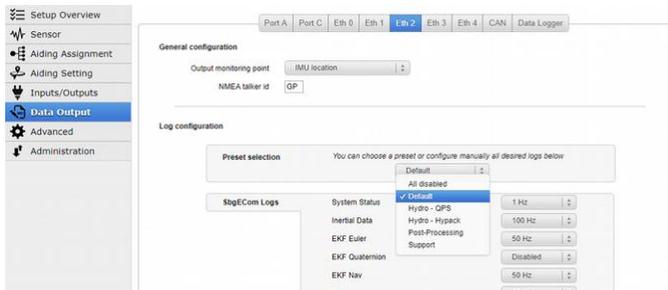
Roll/Pitch (°) Heading (°)

Velocity (m/s) Position (m)

Miscellaneous

Outputting data to third party software

It is possible to select a preset in order to configure the output protocols depending on what is expected by the third party software. For example, if using QPS, Hypack or PDS2000, the Hydro presets will correspond to the SBG Systems driver on the third party software side.

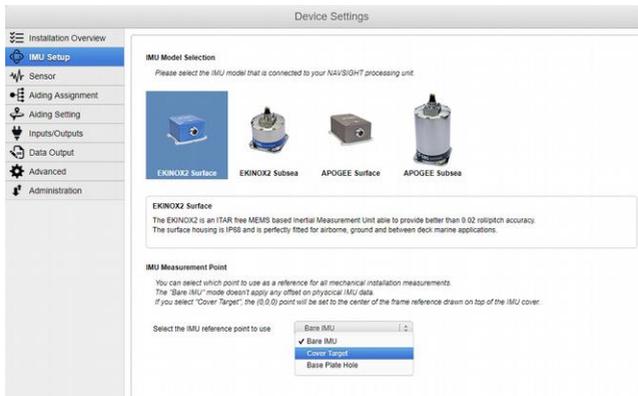


Note: If using a serial port to output data, it is important to make sure the baudrate is high enough for the amount of data being output to avoid saturating the port.

IMU Measurement Point

It is possible to choose what will be the reference point for the IMU for all lever arms measurements. This reference point can be the center of the IMU as described in the drawings in the Hardware Manual, or the target on the top of the housing, or the base plate hole.

Navsight/Quanta IMU measurement point:



Ekinox/Apogee IMU measurement point:

